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## PAPER

## Non-interferometric test of the continuous spontaneous localization model based on rotational optomechanics

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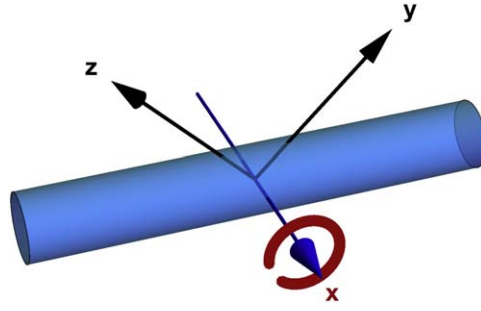
## Abstract

The continuous spontaneous localization (CSL) model is the best known and studied among collapse models, which modify quantum mechanics and identify the fundamental reasons behind the unobservability of quantum superpositions at the macroscopic scale. Albeit several tests were performed during the last decade, up to date the CSL parameter space still exhibits a vast unexplored region. Here, we study and propose an unattempted non-interferometric test aimed to fill this gap. We show that the angular momentum diffusion predicted by CSL heavily constrains the parametric values of the model when applied to a macroscopic object.

## 1. Introduction

Collapse models are widely accepted as a well-motivated challenge to the quantum superposition principle of quantum mechanics. They modify the Schrödinger equation by adding nonlinear and stochastic terms whose action is negligible on microscopic systems, hence preserving their quantum properties, but gets increasingly stronger on macroscopic ones, inducing a rapid collapse of the wave-function in space [1–5]. The most studied and used collapse model is the continuous spontaneous localization (CSL) model. It is characterized by a coupling rate  $\lambda$  between the system and the noise field allegedly responsible for the collapse, and a typical correlation length  $r_C$  for the latter. Ghirardi, Rimini and Weber (GRW) originally set [1]  $\lambda = 10^{-16} \text{ s}^{-1}$  and  $r_C = 10^{-7} \text{ m}$ . Later, Adler suggested different values [6, 7] namely  $r_C = 10^{-7} \text{ m}$  with  $\lambda = 10^{-8 \pm 2} \text{ s}^{-1}$  and  $r_C = 10^{-6} \text{ m}$  with  $\lambda = 10^{-6 \pm 2} \text{ s}^{-1}$ . This shows that there is no consensus so far on the actual values of the parameters.

As the CSL model is phenomenological, the values of  $\lambda$  and  $r_C$  must be eventually determined by experiments. By now there is a large literature on the subject. Such experiments are important because any test of collapse models is a test of the quantum superposition principle. In this respect, experiments can be grouped in two classes: interferometric tests and non-interferometric ones. The first class includes those experiments, which directly create and detect quantum superpositions of the center of mass of massive systems. Examples of this type are molecular interferometry [8–11] and entanglement experiment with diamonds [12, 13]. Actually, the strongest bounds on the CSL parameters come from the second class of non-interferometric experiments, which are sensitive to small position displacements and detect CSL-induced diffusion in position [14–16]. Among them, measurements of spontaneous x-ray emission gives the strongest bound on  $\lambda$  for  $r_C < 10^{-6} \text{ m}$  [17, 18], while force noise measurements on nanomechanical cantilevers [19–21] and on gravitational wave detectors give the strongest bound for  $r_C > 10^{-6} \text{ m}$  [22, 23].



**Figure 1.** Graphical representation of the cylinder with respect to the chosen Cartesian axes. The monitored motions are the vibration along the  $x$ -axis and the rotation around it (represented in blue and red, respectively).

So far research mainly focused on CSL-induced linear diffusion. Very recent technological developments allow to achieve better and better control of rotational motion of non-spherical objects [24–26], thus paving the way to testing rotational CSL-induced diffusion [27–29].

By taking the non-interferometric perspective, in this paper we address the potential effects of the CSL mechanisms on an optomechanical system endowed with heterogeneous degrees of freedom. In particular, we consider the roto-vibrational motion of a system coupled to the field of an optical cavity. By addressing its ensuing dynamics, we show that the rotational degree of freedom offers enhanced possibilities for exploring a wide-spread region of the parameters space of the CSL model, thus contributing significantly to the ongoing quest for the validity of collapse theories. We provide a thorough assessment of the experimental requirements for the envisaged test to be realized and highlight the closeness of our proposal to state-of-the-art experiments.

## 2. Theory

In order to fix the ideas, here we focus on an optomechanical setup whose vibrational and rotational degrees of freedom are monitored. Although the range of masses spanned by typical optomechanical experiments is very large [30] (from the  $zg$  scale of atomic gases [31] to the 40 kg of the LIGO mirrors [32]), the measurement technique is conceptually the same for all cases and it is commonly performed by means of the coupling to an optical mode, which is then readout to infer the noise properties of the mechanical system. Specifically, the density noise spectrum (DNS) of suitable observables of the optical mode is typically used as the workhorse to gather insight into the motion of the mechanical system [33]. As illustrated later on in this section, this will embody our detection scheme for the roto-vibrational motion that we focus on.

### 2.1. Optomechanical setup

Let us consider a cylinder harmonically trapped, both in position and in angle, whose monitored motions are the center of mass vibrations along the  $x$ -axis and the rotations around it. We consider the system symmetry axis oriented orthogonally to the direction of light propagation (see figure 1). The Hamiltonian, describing the vibrational motion of the cylinder harmonically trapped at frequency  $\omega_m$  by interacting with a cavity field, is given by [34, 35]

$$\hat{H}_V = \hbar\omega_C \hat{a}^\dagger \hat{a} + \frac{\hat{p}^2}{2m} + \frac{1}{2}m\omega_m^2 \hat{x}^2 - \hbar\chi \hat{a}^\dagger \hat{a} \hat{x}. \quad (1)$$

In equation (1) the first term describes the free evolution of the cavity mode at frequency  $\omega_C$ , with  $\hat{a}^\dagger$  and  $\hat{a}$  denoting the photons' creation and annihilation operators; the next two terms describe the oscillatory motion of the mass  $m$  in the cavity, where  $\hat{x}$  and  $\hat{p}$  are, respectively, its position and momentum operators. The last term describes the interaction between the cavity field and the vibrational motion of the system, with coupling constant  $\chi$ . If we include also rotations of the cylinder along the direction of propagation of the radiation field ( $x$ -axis) we need to consider the additional term [36, 37]

$$\hat{H}_R = \frac{\hat{L}_x^2}{2I} + \frac{1}{2}I\omega_\phi^2 \hat{\phi}^2 - \hbar g_\phi \hat{a}^\dagger \hat{a} \hat{\phi}, \quad (2)$$

which is the rotational Hamiltonian, characterized by a moment of inertia  $I$  and torsional trapping frequency  $\omega_\phi$ ; the third term, proportional to the coupling constant  $g_\phi$ , accounts for the laser interaction with the rotational degrees of freedom. In equation (2),  $\hat{\phi}$  is the angular operator describing rotations along the  $x$ -axis, such that  $[\hat{\phi}, \hat{L}_x] = i\hbar$ , with  $\hat{L}_x$  the angular momentum operator along the same direction.

## 2.2. CSL model

The master equation describing the evolution of the density matrix of a system affected by a CSL-like mechanism [3, 4] is of the Lindblad form  $\partial_t \hat{\rho} = -\frac{i}{\hbar} [\hat{H}, \hat{\rho}] + \mathcal{L}[\hat{\rho}]$ , where  $\hat{H}$  is the free Hamiltonian of the system and

$$\mathcal{L}[\hat{\rho}] = -\frac{\lambda}{2r_C^3 \pi^{3/2} m_0^2} \int d^3\mathbf{r} [\hat{M}(\mathbf{r}), [\hat{M}(\mathbf{r}), \hat{\rho}]], \quad (3)$$

with  $\hat{M}(\mathbf{r}) = \sum_n m_n \exp(-(\mathbf{r} - \hat{\mathbf{r}}_n)^2 / 2r_C^2)$  and  $\hat{\mathbf{r}}_n$  the position operator of the  $n$ th nucleon (of mass  $m_n$ ) of the system. Under the approximation of small fluctuations of the center-of-mass of a rigid object and small rotations of the system under the action of the CSL noise, two conditions that are fulfilled in typical optomechanical setups,  $\mathcal{L}[\hat{\rho}]$  can be Taylor expanded around the equilibrium position. The center of mass motion and the system's rotations can be decoupled from the internal dynamics and equation (3) reduces to [29]

$$\mathcal{L}[\hat{\rho}] \simeq -\frac{\eta_V}{2} [\hat{x}, [\hat{x}, \hat{\rho}]] - \frac{\eta_R}{2} [\hat{\phi}, [\hat{\phi}, \hat{\rho}]], \quad (4)$$

which represents an extension of the master equation describing only the pure center of mass vibrations (the first of the two terms) to the roto-vibrational case; its general form for an arbitrary geometry of the system can be found in [29]. The explicit forms of the vibrational ( $\eta_V$ ) and rotations ( $\eta_R$ ) diffusion constants are reported in appendix A.

Clearly, equation (4) predicts a diffusion of the linear and angular momentum and optomechanical setups are ideal sensors to measure such effects. There is a large variety employed in these experiments and external influences can be monitored very accurately.

The corresponding equations of motion can be obtained by merging the Hamiltonian optomechanical dynamics in equations (1) and (2) and the CSL-induced diffusions described by equation (4), to which we add the dampings and thermal noises [38]. Explicitly, we get the equations [36]:  $d\hat{x}/dt = \hat{p}/m$ ,  $d\hat{\phi}/dt = \hat{L}_x/I$  and

$$\begin{aligned} \frac{d\hat{a}}{dt} &= -i(\Delta_0 - g_\phi \hat{\phi} - i\kappa) \hat{a} + i\chi \hat{a} \hat{x} + \sqrt{2\kappa} \hat{a}_{\text{in}}, \\ \frac{d\hat{p}}{dt} &= -m\omega_m^2 \hat{x} + \hbar\chi \hat{a}^\dagger \hat{a} - \gamma_m \hat{p} + \hat{\xi}^V - \hbar\sqrt{\eta_V} w_V, \\ \frac{d\hat{L}_x}{dt} &= -I\omega_\phi^2 \hat{\phi} + \hbar g_\phi \hat{a}^\dagger \hat{a} - \frac{D_\phi}{I} \hat{L}_x + \hat{\xi}^R - \hbar\sqrt{\eta_R} w_R, \end{aligned} \quad (5)$$

where  $\Delta_0 = \omega_C - \omega_0$  is the detuning of the laser frequency  $\omega_0$  from the cavity resonance;  $\kappa$ ,  $\gamma_m$  and  $D_\phi$  are the damping rates for the cavity, for the vibrations and rotations of the system, respectively;  $\hat{a}_{\text{in}}$  is a noise operator describing the incident laser field, defined by the input power  $P_{\text{in}} = \hbar\omega_C |\alpha|^2$ , with  $\alpha = \langle \hat{a}_{\text{in}} \rangle$ , and delta-correlated fluctuations  $\langle \delta \hat{a}_{\text{in}}(t) \delta \hat{a}_{\text{in}}^\dagger(s) \rangle = \delta(t-s)$ , where  $\hat{a}_{\text{in}} = \alpha + \delta \hat{a}_{\text{in}}$ . The noise operators  $\hat{\xi}^V$  and  $\hat{\xi}^R$  describe the thermal action of the surrounding environment (supposed to be in equilibrium at temperature  $T$ ), which is assumed to act independently on vibrations and rotations. They are assumed to be Gaussian with zero mean and correlation function [36]

$$\frac{\langle \hat{\xi}_t^j \hat{\xi}_s^j \rangle}{\hbar \epsilon_j} = \int \frac{d\omega}{2\pi} e^{-i\omega(t-s)} \omega [1 + \coth(\beta\omega)] \quad (j = R, V), \quad (6)$$

with  $\beta = \hbar/2k_B T$ ,  $\epsilon_R = D_\phi$ , and  $\epsilon_V = m\gamma_m$ . As already discussed, the CSL noise acts as a source of stochastic noise, whose influence on the dynamics of the system is encompassed by the addition, in equation (5), of the force terms  $-\hbar\sqrt{\eta_j} w_j$  ( $j = R, V$ ) with  $\langle w_j \rangle = 0$  and  $\langle w_i(t) w_j(s) \rangle = \delta_{ij} \delta(t-s)$  [14, 15].

From equations (5) and (6) we can derive the DNS associated to  $\delta \hat{x}(\omega)$  and  $\delta \hat{\phi}(\omega)$ , which are the fluctuations of the position and angle operators in Fourier space, respectively,  $\mathcal{S}_j(\omega) = \frac{1}{4\pi} \int d\Omega \langle \{ \delta \hat{O}_j(\omega), \delta \hat{O}_j(\Omega) \} \rangle$  ( $j = V, R$ ).

Through a lengthy but straightforward calculation, the explicit form of both  $\mathcal{S}_V(\omega)$  and  $\mathcal{S}_R(\omega)$  can be calculated and put under the Lorentzian form

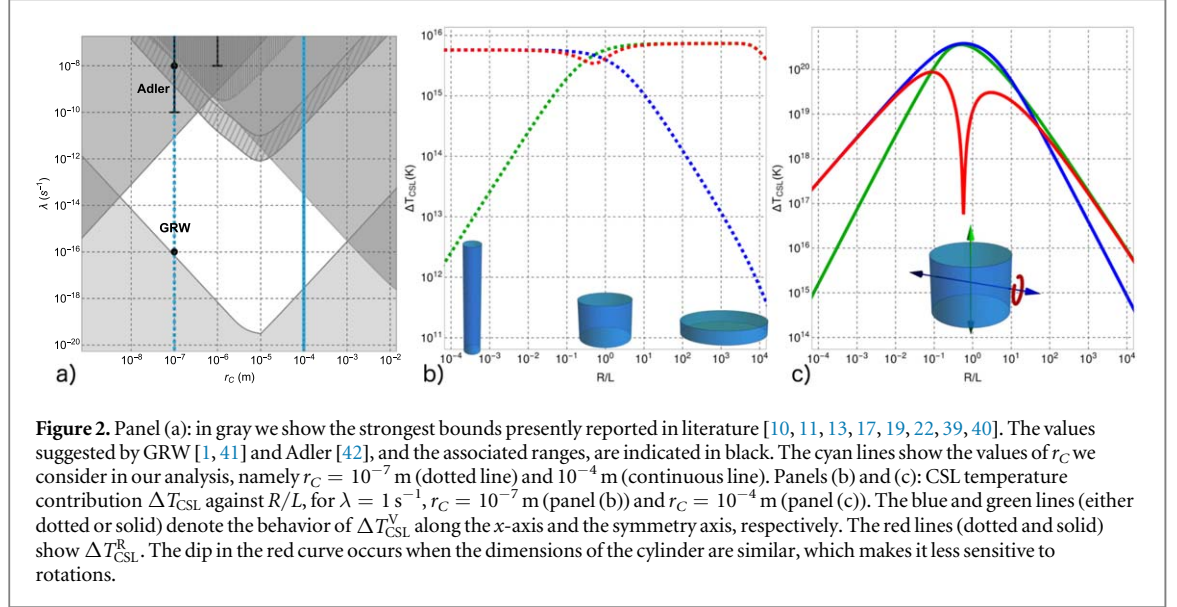
$$\mathcal{S}_j(\omega) = \frac{2\hbar^2 |\alpha|^2 \kappa \mathcal{G}_j^2 + [\kappa^2 + (\Delta - \omega)^2] [\hbar\omega \epsilon_j \coth(\beta\omega) + \hbar^2 \eta_j]}{\lambda_j^2 [\kappa^2 + (\Delta - \omega)^2] [(\omega_{j,\text{eff}}^2 - \omega^2)^2 + \Gamma_{j,\text{eff}}^2 \omega^2]}. \quad (7)$$

The parameters specific of the considered degree of freedom that appear in equation (7) ( $\mathcal{G}_j$ ,  $\omega_{j,\text{eff}}$ ,  $\Gamma_{j,\text{eff}}$  and  $\lambda_j$ ) are given in table 1.

We have introduced the effective frequencies  $\omega_{m,\text{eff}}$  and  $\omega_{\phi,\text{eff}}$  and damping constants  $\gamma_{m,\text{eff}}$  and  $D_{\phi,\text{eff}}$  [36, 43], whose explicit expressions are presented in appendix B, and  $\Delta = \Delta_0 - g_\phi \langle \hat{\phi} \rangle - \chi \langle \hat{x} \rangle$ . The CSL contributions are encompassed by the diffusion constant  $\eta_j$ , which enters  $\mathcal{S}_j(\omega)$  as an additional heating term akin to the environment-induced one  $\hbar\omega \epsilon_j \coth(\beta\omega)$ . In the high temperature limit ( $\beta \rightarrow 0$ ), which is in general valid for typical low-frequency optomechanical experiments, the latter takes the form  $\hbar\epsilon_j/\beta$ . Therefore, in such a

**Table 1.** Explicit form of the parameters entering the DNS of the fluctuations of the rotational and vibrational degrees of freedom of the system (see equation (7)).

DNS parameter	$\mathcal{G}_j$	$\omega_{j,\text{eff}}$	$\Gamma_{j,\text{eff}}$	$\lambda_j$
Vibration	$\chi$	$\omega_{\text{m,eff}}$	$\gamma_{\text{m,eff}}$	$m$
Rotation	$g_\phi$	$\omega_{\phi,\text{eff}}$	$D_{\phi,\text{eff}}/I$	$I$



limit, we have

$$\hbar\omega\epsilon_j \coth(\beta\omega) + \hbar^2\eta_j \rightarrow \hbar\epsilon_j \left( \frac{1}{\beta} + \frac{\hbar\eta_j}{\epsilon_j} \right) \equiv \frac{\hbar\epsilon_j}{\beta_{j,\text{eff}}}, \quad (8)$$

where we have defined the  $j$ -dependent effective inverse temperature  $\beta_{j,\text{eff}}$ , thus showing that the different degrees of freedom of the system thermalize to different, in principle distinguishable, CSL-determined temperatures. This means that CSL gives the extra temperatures

$$\Delta T_{\text{CSL}}^V = \frac{\hbar^2\eta_V}{2k_B m \gamma_m} \quad \text{and} \quad \Delta T_{\text{CSL}}^R = \frac{\hbar^2\eta_R}{2k_B D_\phi}. \quad (9)$$

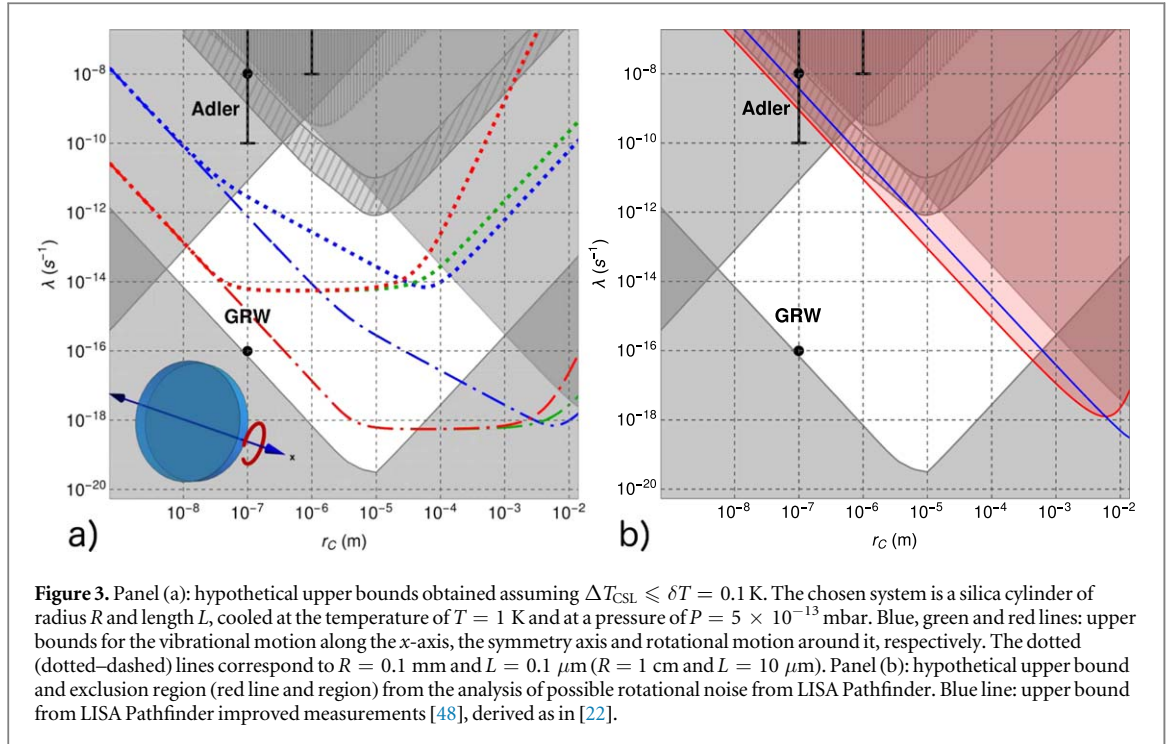
The first was extensively studied both theoretically [14, 15, 27, 44–46] and experimentally [13, 19, 22]. However, for the rotational degree of freedom, the existence of  $\Delta T_{\text{CSL}}^R$  opens up new possibilities for testing the CSL model, as discussed below.

### 3. Lab-based experiments

Upon subtracting the optomechanical contribution to the temperature embodied by the first term in equation (7)<sup>7</sup>, the experimental measurement of the temperature of the system is given by  $T_m \pm \delta T$ , where  $\delta T$  is the experimental measurement accuracy. Unless one sees an excess temperature of unknown origin [20], the outcome of the experiment will be  $\Delta T_{\text{CSL}} \leq \delta T$ , thus setting a bound on the collapse parameters once equation (9) is considered. We first compare the magnitude of the two temperatures  $\Delta T_{\text{CSL}}^V$  and  $\Delta T_{\text{CSL}}^R$  for different geometries of the system. Without loss of generality (as  $\Delta T_{\text{CSL}}^{V,R} \propto \eta_j \propto \lambda$ ) we set  $\lambda = 1 \text{ s}^{-1}$ . For definiteness we take a silica cylinder with  $m = 10 \text{ }\mu\text{g}$  and vary the ratio between the radius  $R$  and the length  $L$ . For the residual gas, we consider He-4, at the temperature of  $T = 1 \text{ K}$  and pressure  $P = 5 \times 10^{-13} \text{ mbar}$ , which can be reached with existing technology [47]. Figure 2 shows the behavior of  $\Delta T_{\text{CSL}}^{V,R}$  for  $r_c = 10^{-7}$  m (dotted

<sup>7</sup> The contribution arising from the driving field can be accurately calibrated experimentally due to the relatively large intensity of the field (which also makes any uncertainty negligible with respect to the nominal signal). Such well characterized contribution can then be subtracted from the DNS to let the features linked to the mechanical motion emerge. We also note that the optomechanical contribution can be strongly suppressed by implementing a stroboscopic measurement strategy.





lines) and  $r_C = 10^{-4}$  m (continuous lines)<sup>8</sup>. In the latter case the strongest contribution comes from vibrations along the  $x$ -axis (blue lines) at  $2R \sim L$ , while in the former case it comes from rotations (red lines) for  $R \gg L$  and  $R \ll L$ , thus showing that CSL tests based on rotational motion can be as good or better than those based on vibrational motion. For the following analysis we focus on the  $R \gg L$  case, which gives the strongest contribution for the originally chosen value of the correlation length  $r_C = 10^{-7}$  m. As a comparison, we also report  $\Delta T_{\text{CSL}}$  given by vibrations along the symmetry axis (green lines).

In figure 3(a) we compare the hypothetical upper bounds obtained from the vibrational and rotational motion, taken individually. This is done by setting the accuracy in temperature to  $\delta T = 0.1$  K and varying the dimensions of the cylinder. As a case-study, we consider a thought experiment aimed at testing CSL in the region  $r_C \sim 10^{-7}$  m, and exploit rotations of a *coin* shaped system to maximize the CSL effect (see figure 2(b)). As shown, the hypothetical upper bound given by the rotational motion is comparable with the vibrational one.

### 3.1. Experimental feasibility

Having assessed formally how, in extreme vacuum conditions, the rotational motion of a levitated cylinder of fairly macroscopic dimensions can set very strong bounds on the CSL model (almost testing the GRW hypothesis), we now address the experimental feasibility of our proposal, showing that the proposed experiment is entirely within the grasp of current technology.

First, the cylinder has to be trapped magnetically or electrically to allow for its rotational motion around the  $x$ -axis, as in figure 1. Needless to say, we must avoid competing heating effects such those due to gas collisions and exchange of thermal photons between the environment and the trapped cylinder. Moreover, one must ensure the ability to control and detect precisely enough the rotational motion of the trapped cylinder. The first condition can be granted by performing the experiments at low temperatures and pressures. Standard dilution cryostats reach temperatures  $< 10$  mK [19, 20] and the reachable pressure is as low as  $10^{-17}$  mbar (as done in the cryogenic Penning-trap experiment reported in [47]), which is much lower than the  $5 \times 10^{-13}$  mbar considered in figures 3(a), (b). The trapping of the cylinder can be done magnetically or electrically [49, 50], while the control and readout of the rotational motion can be achieved by an optical scattering technique [51].

Further, a stroboscopic detection mode can be chosen to suppress the heating by the detection light of the rotational motion of the cylinder [46]. The rotational state can be prepared very reliably by feedback control [52]. The feedback is then turned off to allow for heating.

Alternatively, if a magnetic cylinder is levitated and trapped, SQUID sensors could be used to read the rotational state. The most notable advantage of this approach is that levitation can be achieved with static fields, implying negligible heat leak. Moreover, a temperature resolution better than 0.1 K has already been demonstrated for state-of-the-art high quality cantilevers with a SQUID-based magnetic detection [19, 53].

<sup>8</sup> These values of  $r_C$  are chosen due to their closeness to the boundaries of the unexplored CSL parameter region.

Therefore the main requirements for this proposal can be reached in a dedicated experiment based on existing technology.

Clearly a big experimental challenge will be the control of seismic and acoustic noise and other environmental effects [54]. In this respect, rotational degrees of freedom can be decoupled from vibrational noise much more effectively than vibrational ones. A well known and paradigmatic example is given by the torsion pendulum, which is by far the most effective method to measure forces at Hz and sub-Hz frequencies.

Notice also that this non-interferometric test does not require the preparation of any non-classical state, which would need much advanced technology, yet to be demonstrated for such a macroscopic object. While the experimental scenario and the shape of the cylinder is the same as discussed already in [27, 28], here we find that macroscopic dimensions for the cylinder are useful for testing collapse models. This should make our proposed test far less demanding than experimenting with a nano- or micro-scale cylinder.

#### 4. Space-based experiments

An example of experiment in which rotational measurements can in principle improve the bounds on CSL with respect to translational tests is the space mission LISA Pathfinder, whose preliminary data for the vibrational noise were already exploited to set upper bounds on the CSL parameters [22, 23]. We can readily apply our model to LISA Pathfinder, with minimal variations to take into account the cubic instead of cylindrical geometry. The core of the experiment consists in a pair of test masses (TMs) in free-fall, surrounded by a satellite which follows the masses while minimizing the stray disturbances. As the satellite does not rigidly move with the masses, there is the necessity of setting a reference. One thus considers two masses in place of a single one, and focus on their relative motion. The geometry of each TM is that of a cube of side  $L = 4.6$  cm and mass  $m = 1.928$  kg made of an AuPt alloy. The distance between the masses is  $a = 37.6$  cm. Under the LISA Pathfinder conditions and provided that the noise is dominated by gas damping, in the limit of  $r_C/L \rightarrow 0$ , we find that the torque over the force DNS ratio is 4 times bigger for the CSL noise than for the residual gas noise, showing that is advantageous to set bounds on CSL by looking at rotational noise. Though the data from the rotational noise measurements are not yet available, we can set an hypothetical bound on CSL parameters by converting the force DNS  $S_F(\omega) = 3.15 \times 10^{-30} \text{ N}^2 \text{ Hz}^{-1}$  [48] in torque DNS  $S_{\tau_x}(\omega) = 0.04 \times S_F(\omega)L^2 = 2.66 \times 10^{-34} \text{ N}^2 \text{ m}^2 \text{ Hz}^{-1}$ . Such a value is compared with the CSL contribution  $S_{\tau_x}(\omega) = \frac{1}{2}2\hbar^2\eta_R^{(\text{cube})}$ , where  $\eta_R^{(\text{cube})}$  is given in equation (A3) and the factors  $\frac{1}{2}$  and 2 account, respectively for the differential measurement of the two masses of LISA Pathfinder and for the conversion from the two-side to one-side spectra. The corresponding upper bound and excluded region are shown in red in figure 3(c). In blue, we report the upper bound from the vibrational analyses performed in [22] with the improved data from [48]. The rotational upper bound would correspond to an enhancement of a factor 4 with respect to the improved vibrational bound, which is already almost one order of magnitude stronger than the one previously established [22]. This shows that a rotation-based tests hold the potential to refine the probing of CSL mechanisms.

We underline that this bound is hypothetical, as the rotational noise is theoretically estimated. The measurement accuracy of the rotational motion is expected to be worse since the interferometric measurement of LISA Pathfinder is optimized for the vibrational degrees of freedom [55]. To get a rotational bound, which is stronger than the vibrational one, one needs  $S_{\tau_x}(\omega) = 1.07 \times 10^{-33} \text{ N}^2 \text{ m}^2 \text{ Hz}^{-1}$ . This should be within reach of the LISA Pathfinder technology. A more technical discussion that includes environmental noise is given in appendix C.

A final note: the hypothetical rotational upper bound would completely rule out the possibility that the excess noise measured in the improved cantilever experiment [20] is due to the CSL noise.

#### 5. Conclusions

The CSL parameter space has been the focus of a growing number of theoretical and experimental investigations aimed at reducing it significantly. To date, the region of the parameter space for this model that has not been excluded explicitly is still many orders of magnitude wide both in the values of the correlation length  $r_C$  and the localization rate  $\lambda$  (see white region in figure 2(a)). We have proposed a non-interferometric test capable of probing such a region. The difference with the tests that have already been suggested and performed is twofold. First, our proposal is built on the use of rotational degrees of freedom rather than the usual vibrational ones. Second, the scheme focuses on objects of macroscopic dimensions instead of micro-scale ones. As discussed in detail in this work, both aspects offer considerable advantages that were at the basis of the reduction of the parameter space mentioned above. Although both features above have already been discussed and studied



individually, an investigation combining such advantages together is unique of our proposal. We believe that the test that has been put forward here, which has been shown to adhere well to the current experimental state-of-the-art, provides a new avenue of great potential for testing the CSL model.

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## Appendix A. CSL Diffusion coefficients

The CSL diffusion coefficients have been already computed in [14, 15, 29]. Given the mass density  $\mu(\mathbf{r})$ , they read

$$\begin{aligned}\eta_V &= \frac{\lambda r_C^3}{\pi^{3/2} m_0^2} \int d^3\mathbf{k} e^{-r_C^2 k^2} k_x^2 |\tilde{\mu}(\mathbf{k})|^2, \\ \eta_R &= \frac{\lambda r_C^3}{\pi^{3/2} m_0^2} \int d^3\mathbf{k} e^{-r_C^2 k^2} |k_y \partial_{k_z} \tilde{\mu}(\mathbf{k}) - k_z \partial_{k_y} \tilde{\mu}(\mathbf{k})|^2,\end{aligned}\quad (\text{A1})$$

where  $\tilde{\mu}(\mathbf{k})$  is the Fourier transform of  $\mu(\mathbf{r})$ . For a cylinder of length  $L$  and radius  $R$  we have [15, 29]

$$\begin{aligned}\eta_V^{(\text{cyl})} &= \frac{8m^2 r_C^2 \lambda}{L^2 m_0^2 R^2} I_1\left(\frac{R^2}{2r_C^2}\right) e^{-\frac{R^2}{2r_C^2}} \left( \frac{L\sqrt{\pi}}{2r_C} \operatorname{erf}\left(\frac{L}{2r_C}\right) - 1 + e^{-\frac{L^2}{4r_C^2}} \right), \\ \eta_V^{(\text{cyl}, \text{sym})} &= \frac{8m^2 r_C^2 \lambda}{L^2 m_0^2 R^2} \left( 1 - e^{-\frac{L^2}{4r_C^2}} \right) \left( 1 - e^{-\frac{R^2}{2r_C^2}} \left( I_0\left(\frac{R^2}{2r_C^2}\right) + I_1\left(\frac{R^2}{2r_C^2}\right) \right) \right), \\ \eta_R^{(\text{cyl})} &= \frac{2\lambda r_C^4 m^2}{L^2 R^2 m_0^2} \left\{ \left[ \left( 1 - e^{-\frac{L^2}{4r_C^2}} \right) \left( 8 + \frac{R^2}{r_C^2} \right) - \frac{2L\sqrt{\pi}}{r_C} \operatorname{erf}\left(\frac{L}{2r_C}\right) \right] \right. \\ &\quad \left. - 2I_0\left(\frac{R^2}{2r_C^2}\right) e^{-\frac{R^2}{2r_C^2}} \left[ \left( 1 - e^{-\frac{L^2}{4r_C^2}} \right) \left( 4 + \frac{R^2}{r_C^2} \right) - \frac{L\sqrt{\pi}}{r_C} \operatorname{erf}\left(\frac{L}{2r_C}\right) \right] \right. \\ &\quad \left. - \frac{1}{3} I_1\left(\frac{R^2}{2r_C^2}\right) e^{-\frac{R^2}{2r_C^2}} \left[ \left( 3 - e^{-\frac{L^2}{4r_C^2}} \right) \frac{L^2}{r_C^2} + 2 \left( 1 - e^{-\frac{L^2}{4r_C^2}} \right) \left( 14 + \frac{3R^2}{r_C^2} \right) - \frac{L\sqrt{\pi}}{2r_C} \left( 24 + \frac{L^2}{r_C^2} \right) \operatorname{erf}\left(\frac{L}{2r_C}\right) \right] \right\},\end{aligned}\quad (\text{A2})$$

where  $I_n$  denotes the  $n$ th modified Bessel function. We also need the following coefficient

$$\begin{aligned}\eta_R^{(\text{cube})} &= \frac{8\lambda}{3} \left( \frac{m}{m_0} \right)^2 \left( \frac{r_C}{L} \right)^6 \left( 1 - e^{-\frac{L^2}{4r_C^2}} - \frac{\sqrt{\pi} L}{2r_C} \operatorname{erf}\left(\frac{L}{2r_C}\right) \right) \\ &\quad \times \left\{ \left( 1 - e^{-\frac{L^2}{4r_C^2}} \right) \left[ 2 \left( 3 - e^{-\frac{L^2}{4r_C^2}} \right) \left( \frac{L}{r_C} \right)^2 + 32 \left( 1 - e^{-\frac{L^2}{4r_C^2}} \right) \right. \right. \\ &\quad \left. \left. - \frac{\sqrt{\pi} L}{r_C} \left[ 24 + \left( \frac{L}{r_C} \right)^2 \right] \operatorname{erf}\left(\frac{L}{2r_C}\right) \right] + 3\pi \left( \frac{L}{r_C} \right)^2 \operatorname{erf}^2\left(\frac{L}{2r_C}\right) \right\},\end{aligned}\quad (\text{A3})$$

which refers to a cube of side  $L$ .

## Appendix B. Effective frequencies and damping constants

The effective frequencies  $\omega_{\text{m,eff}}$  and  $\omega_{\phi,\text{eff}}$  and damping constants  $\gamma_{\text{m,eff}}$  and  $D_{\phi,\text{eff}}$  introduced in table 1 take the following form

$$\begin{aligned}\omega_{\text{m,eff}}^2 &= \omega_{\text{m}}^2 - \frac{2\hbar\chi^2|\alpha|^2\Delta(\kappa^2 + \Delta^2 - \omega^2)}{m(\kappa^2 + (\Delta - \omega)^2)(\kappa^2 + (\Delta + \omega)^2)}, \\ \omega_{\phi,\text{eff}}^2 &= \omega_{\phi}^2 - \frac{2\hbar g_{\phi}^2|\alpha|^2\Delta(\kappa^2 + \Delta^2 - \omega^2)}{I(\kappa^2 + (\Delta - \omega)^2)(\kappa^2 + (\Delta + \omega)^2)},\end{aligned}\quad (\text{B1})$$

$$\begin{aligned}\gamma_{\text{m,eff}} &= \gamma_{\text{m}} + \frac{4\hbar\chi^2|\alpha|^2\kappa\Delta}{m(\kappa^2 + (\Delta - \omega)^2)(\kappa^2 + (\Delta + \omega)^2)}, \\ D_{\phi,\text{eff}} &= D_{\phi} + \frac{4\hbar g_{\phi}^2|\alpha|^2\kappa\Delta}{(\kappa^2 + (\Delta - \omega)^2)(\kappa^2 + (\Delta + \omega)^2)}.\end{aligned}\quad (\text{B2})$$

The damping constants  $\gamma_{\text{m}}$  and  $D_{\phi}$  can be expressed in terms of the parameters of the system [56]

$$\begin{aligned}\gamma_{\text{m}} &= \frac{P}{m} \sqrt{\frac{2\pi m_{\text{gas}}}{k_{\text{B}} T}} R^2 \left[ 1 + \frac{3L}{2R} \left( 1 + \frac{\pi}{6} \right) \right], \\ D_{\phi} &= P \sqrt{\frac{\pi m_{\text{gas}}}{2k_{\text{B}} T}} R^4 \left[ 1 + \frac{\pi}{4} + \frac{L}{R} + \frac{1}{2} \left( \frac{L}{R} \right)^2 \right. \\ &\quad \left. + \frac{1}{4} \left( \frac{L}{R} \right)^3 \left( 1 + \frac{\pi}{6} \right) \right],\end{aligned}\quad (\text{B3})$$

where  $P$  is the pressure of the surrounding gas of particles of mass  $m_{\text{gas}}$ . For the vibrational motion along the symmetry axis the damping rate  $\gamma_{\text{m}}$  must be substituted by the following expression [56]

$$\gamma_{\text{m}}^{\text{sym}} = \frac{P}{m} \sqrt{\frac{8\pi m_{\text{gas}}}{k_{\text{B}} T}} R^2 \left( 1 + \frac{\pi}{4} + \frac{L}{2R} \right). \quad (\text{B4})$$

This gives the green lines in figure 2.

## Appendix C. Analysis of LISA Pathfinder noises

Whether one can set stronger bounds on collapse models by looking at the translational or the rotational noise of a given mechanical system depends crucially on the specific experimental implementation. In particular, one has to compare how the CSL noise and the dominant (physical) residual noise scale when passing from translational to rotational noise. If the scaling is different, the bounds that can be inferred from the same experimental setup under the same conditions are different. Here, we show that, for the specific experiment of LISA Pathfinder, under the assumption that residual gas is the dominant source of noise, rotational noise is in principle the best choice. We limit our analysis to the short CSL length limit  $r_{\text{C}} \ll L$ , which is the relevant one in the case of LISA.

We introduce a dimensionless factor  $\alpha$ , defined as:

$$\alpha_i L^2 = \frac{\mathcal{S}_{\tau,i}}{\mathcal{S}_{\text{F},i}}, \quad (\text{C1})$$

where  $\mathcal{S}_{\tau,i}$  and  $\mathcal{S}_{\text{F},i}$  are the torque and force DNS, respectively, and the pedices  $i$  may refer to the three specific cases ‘CSL’, ‘gas- $\infty$ ’ and ‘gas’ which we will now discuss. For the residual gas noise we consider both the case of gas within an infinite volume (gas- $\infty$ ) and the real case of a gas constrained in a small gap  $d \ll L$  (gas), which is the relevant one for LISA Pathfinder. Essentially,  $\alpha$  is the effective ratio of rotational (torque) noise to vibrational (force) noise for a given source.

For CSL in the case of a cubic mass with  $r_{\text{C}}/L \rightarrow 0$ , comparison of equation (A3) with the formula for the vibrational diffusion constant [15, 19]:

$$\eta_{\text{V}}^{(\text{cube})} = \frac{32\lambda r_{\text{C}}^4 m^2}{m_0^2 L^6} \left( \frac{\sqrt{\pi} L}{2r_{\text{C}}} \text{erf}\left(\frac{L}{2r_{\text{C}}}\right) - 1 + e^{-\frac{L^2}{4r_{\text{C}}^2}} \right)^2 \left( 1 - e^{-\frac{L^2}{4r_{\text{C}}^2}} \right), \quad (\text{C2})$$

in the same limit, provides  $\alpha_{\text{CSL}} = 1/6 \simeq 0.166$ . This factor is the same as for a gas of uncorrelated particles scattering elastically off the TM, which is the typical picture considered in elementary textbooks of statistical mechanics. In fact, under elastic scattering the force exerted by the gas is normal to the surface and the total force noise is proportional to the exposed area. One can write  $d\mathcal{S}_{\text{F}} = a \cdot dA$ , and  $d\mathcal{S}_{\tau} = ab^2 \cdot dA$ , where  $a$  is the noise strength and  $b$  is arm of the force, i.e. the distance between the normal to the surface at a given point and the rotational axis. Elementary integration of the force and torque on each cube face provides precisely the factor  $\alpha = 1/6$  regardless of the direction of the force and the torque. In this sense, CSL in the  $r_{\text{C}} \ll L$  limit behaves essentially as a gas of uncorrelated particles hitting the surface normally, which can be interpreted as a collection of uncorrelated collapse events localized on the cube surface.

For a gas in a real experiment, the elastic scattering assumption is known to be wrong. A vast experimental evidence suggests that the data are instead consistent with the inelastic diffuse scattering model [56]. In this picture, a particle hitting the surface with a given angle  $\theta_i$  is reflected with a different angle  $\theta_r$ , with joint probability proportional to  $\cos \theta_i \times \cos \theta_r$  and with uncorrelated incidence and emission velocities consistent with a Maxwell–Boltzmann distribution. In general a shear force component will appear in addition to the normal component. Detailed calculations have been carried out analytically for a gas of particles within an infinite volume [56]. For a cube it is found that the ratio of rotational to vibrational noise is  $\alpha_{\text{gas-}\infty} \simeq 0.226 > \alpha_{\text{CSL}}$ . Therefore, for infinite volume it is not advantageous to set bounds on CSL by looking at rotational noise.

However, the situation of LISA Pathfinder is slightly more complex. The cubic TM is enclosed in an external caging, the gravitational reference sensor (GRS), with a relatively narrow gap between TM and GRS, in the range  $d/L = [0.063, 0.087] \ll 1$  depending on the axis. The gap is narrow in order to enable continuous monitoring (with subsequent control of the GRS) of the relative position between GRS and TM by means of capacitive electrodes.

Under the gap constraint, each individual gas particle inside the gap will undergo a random-walk with a large number of multiple collisions, introducing a degree of correlation between consecutive events. Extensive investigation of this effect has been carried out, based on numerical simulations and experiments with torsion pendulums [57, 58]. The multiple scattering is found to introduce a correlation time  $\tau_c$ , related to the mean time required by a particle to random-walk along the gap from one-side to the other side of a cubic face. At frequencies  $\omega \tau_c \ll 1$ , the relevant case for LISA Pathfinder, there is a significant increase of both the vibrational force and rotational noise, and related damping factors. In the limit  $d \ll L$  the increase goes asymptotically as  $(L/d)^2$ .

It turns out that the increase of vibrational noise is significantly larger than the increase of rotational noise. This can be intuitively understood as following. When a gas particle, during its random-walk, crosses the center of a cubic face, the sign of the torque changes whereas the sign of the force does not. As a consequence, the correlation time for the rotational case is shorter by a factor around 4, being related with the time required to random-walk along half of the cubic face instead of the whole face.

A quantitative estimation can be done by inspection of figure 5 in [57]. Under the LISA Pathfinder condition, the force noise will increase with respect to the infinite volume case by a factor roughly 6 times larger than the torque noise. This leads to  $\alpha_{\text{gas}} \simeq 0.04 < \alpha_{\text{CSL}}$ . Thus, provided that the noise in LISA Pathfinder is dominated by gas damping, rotational noise allows to set an upper bound on CSL which is about 4 times better than using vibrational noise. This is shown in figure 3(c) of the main text.

It is worth mentioning a final aspect. Rotational measurements in LISA Pathfinder are slightly more complicated than vibrational ones. In fact, the residual rotational noise is obtained from a differential measurement scheme which provides a cancellation of the actuation noise [55]. As the latter is two orders of magnitude larger than the former, a very accurate calibration of the actuation noise is needed. On the other hand, an imperfect cancellation of the actuation noise will unavoidably degrade the rotational noise spectrum. This might eventually spoil the advantage of using rotational noise to set an ultimate bound on CSL with LISA Pathfinder. This issue could be solved in a dedicated similar experiment in which the readout is designed to directly measure rotations instead of vibrations.

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